

## ***Numerical Study of Lateral – Torsion Stability of Thin – wall Beams with Cut-out in Web***

B. Poursartip; M. Z. Kabir ;

### ***ABSTRACT***

One of the main challenges in designing of thin-walled open section beams is lateral-torsion buckling mode. Due to the access for piping, electrical and other devices, inserting cut-outs in web are inevitable.

This paper deals with the effects of different parameters in lateral-torsion buckling of plate girders. Lateral-torsion buckling loads of more than 700 plate girders are calculated with opening in different shapes, dimensions and locations by Ansys5.4 . The study is performed using numerical approach and the results are compared together. Three kinds of cut-outs including square, circular and hexagonal are selected. An extensive parametric studies is conducted to study the influence of cut-out dimensions, thickness, area, location and distribution along the beam length. Finally optimum scheme is obtained in order to have better estimation load carrying capacity of open-section beam.

Email: [b\\_poursartip@yahoo.com](mailto:b_poursartip@yahoo.com) ( )

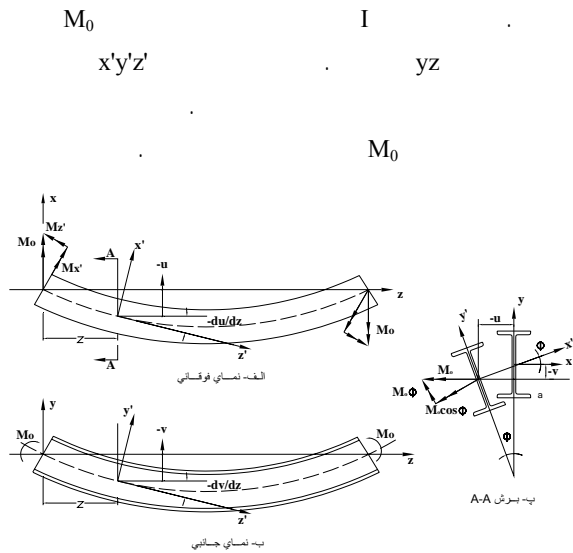
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**KEYWORDS**

Plate girder, cut-out, buckling, web, thin-wall, restrains



Redwood and Uenoya

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A-A

Coul and Alvarez

$M_0$  xyz

$$(M_x)_{ext} = M_0 \cos \varphi \quad ( )$$

$$(M_y)_{ext} = M_0 \sin \varphi = M_0 \quad ( )$$

$$(M_z)_{ext} = -\frac{du}{dz} M_0 \quad ( )$$

$$(M_x)_{int} = -EI_x \frac{d^2 v}{dz^2} \quad ( )$$

$$(M_y)_{int} = EI_y \frac{d^2 u}{dz^2} \quad ( )$$

$$(M_z)_{int} = GJ \frac{d\varphi}{dz} \quad ( )$$

I

v u

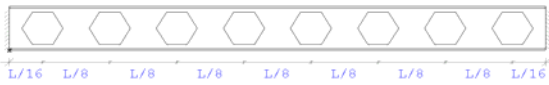
Thevendran and Shanmugam



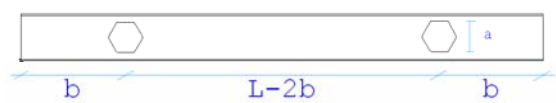
(mm)	( )	( )	( )	( )
I				
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A : ( )



B : ( )



C : ( )

(E)

/ (v) \* kg/m

:[ ]

$$M_{cr} = \sqrt{\frac{\pi^4 E^2 C_w I_y}{L^4} + \frac{\pi^2 E I_y G J}{L^2}} \quad ( )$$

(S<sub>x</sub>)

$$\sigma_{cr} = \sqrt{\frac{\pi^4 E^2 C_w I_y}{S_x^2 L^2} + \frac{\pi^2 E I_y G J}{S_x^2 L^2}} \quad ( )$$

G E σ<sub>cr</sub>  
C<sub>w</sub> y I<sub>y</sub>  
L y S<sub>x</sub> J  
: C<sub>w</sub> J I

$$J = \frac{1}{3} \sum b t^3 \quad ( )$$

$$C_w = \frac{1}{2} I_f h^2 \quad ( )$$

I<sub>f</sub> h y t b

$$C_4 = C_1 \pi \left[ \sqrt{1 + \left(\frac{\pi}{\lambda L}\right)^2} \cdot (C_2^2 + 1) \pm C_2 \frac{\pi}{\lambda L} \right] \quad ( )$$

$$\lambda^2 = \frac{1}{a^2} = \frac{G J}{E C_w} \quad ( )$$

C

C

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Shell

U<sub>x</sub> U<sub>y</sub> U<sub>z</sub> R<sub>y</sub> R<sub>z</sub>

y x z

I	Pa (kg)	Pans (kg)	( )
PI CCR			/
PI CSR			/
PI SCR			/
PI SSR			/
PI CCR			/
PI CSR			/
PI SCR			/
PI SSR			/

Ansys

Ansys

y x  
y  
z  
y x

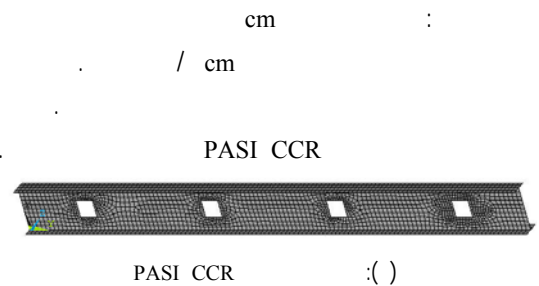
PI CCR  
I . P .  
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CR  
SR

(P<sub>cr</sub>)

(P<sub>cr0</sub>)

(PASI CCR )  
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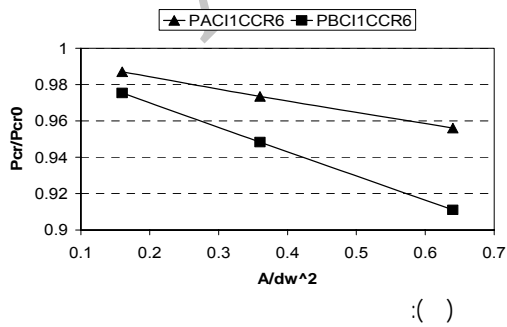
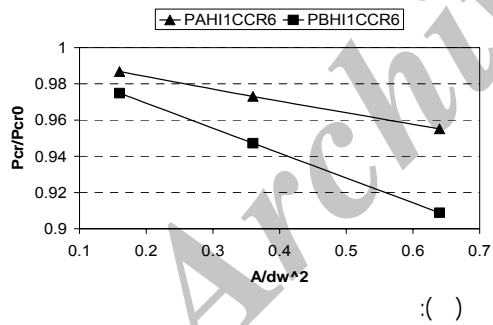
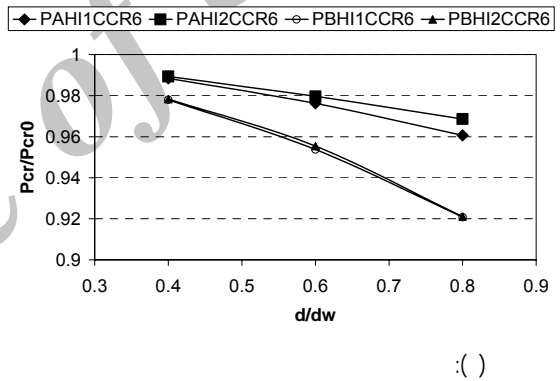
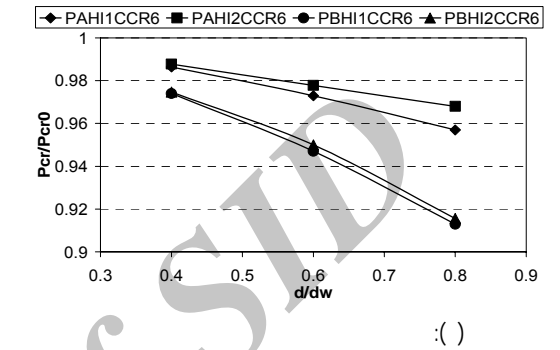
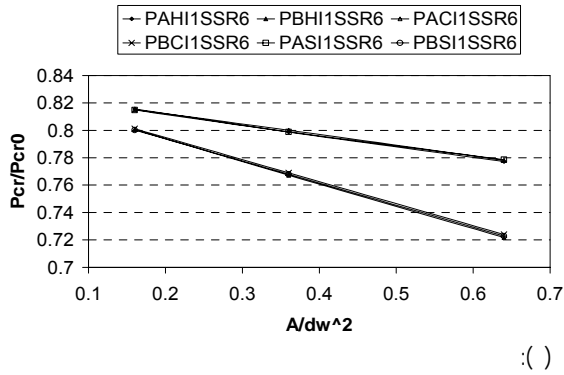
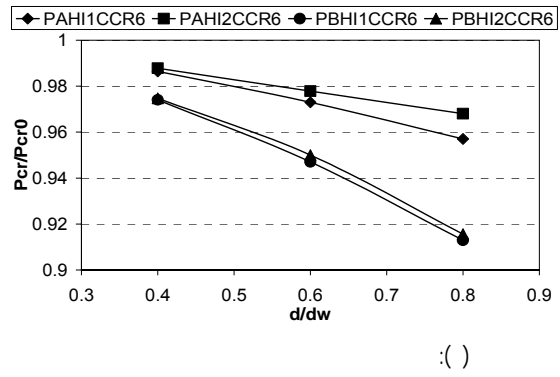
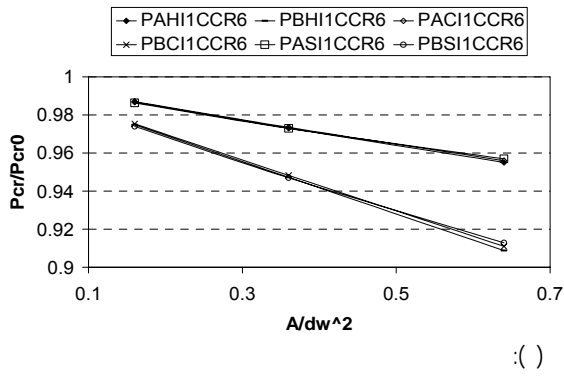
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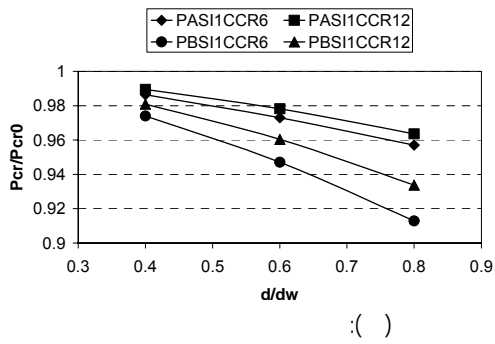
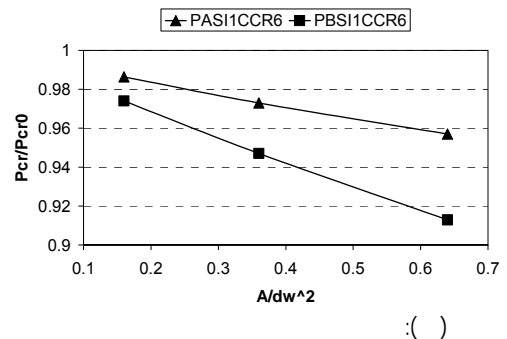
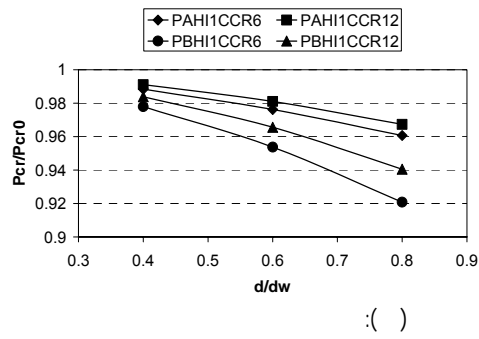
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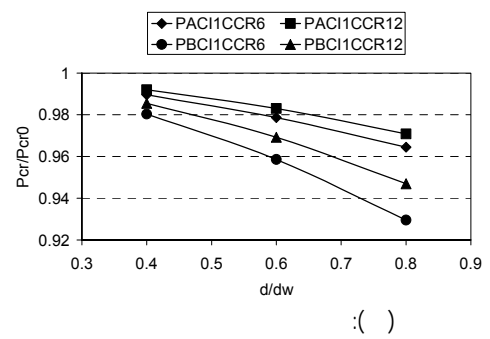
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$P_{cr}/P_{cr0}$

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$a/dw = /$        $b/L = /$        $a/dw = /$   
 $.b/L = /$        $b/L = /$



$a/d_w$

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$L/d_w$

$b/L = /$

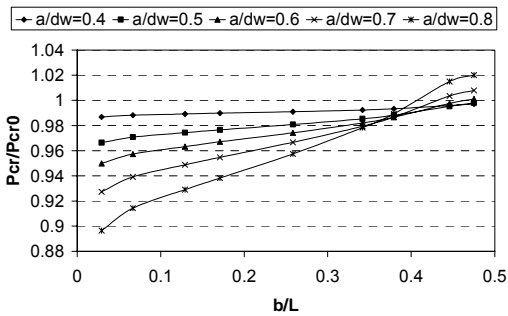
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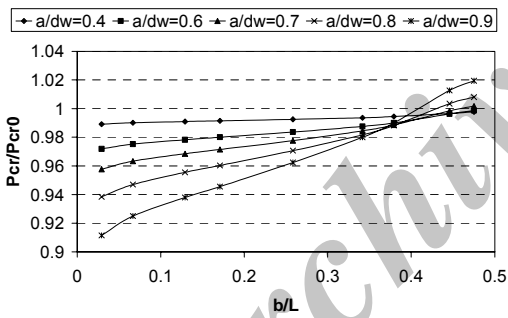
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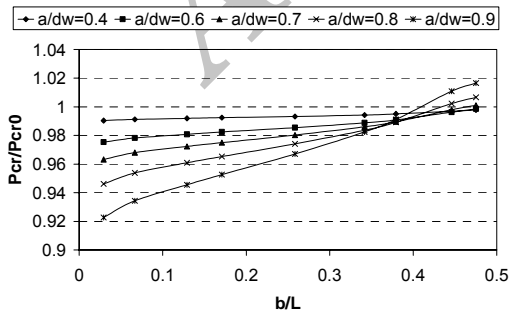
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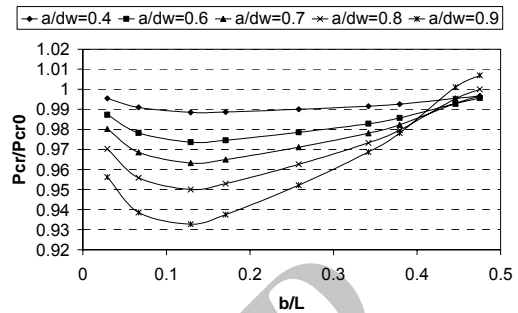
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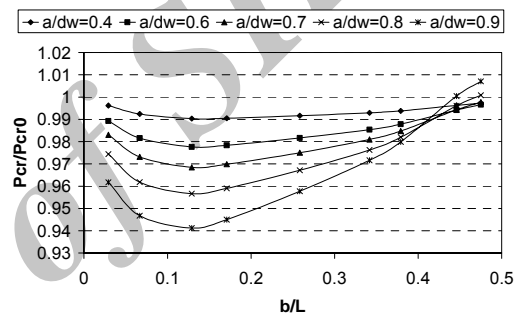
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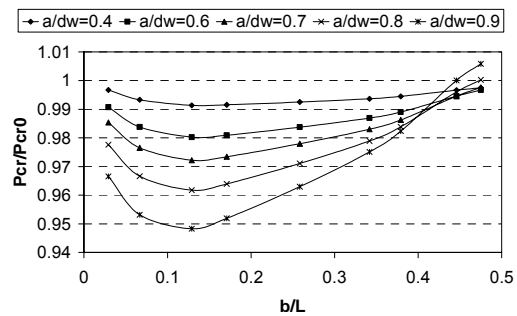
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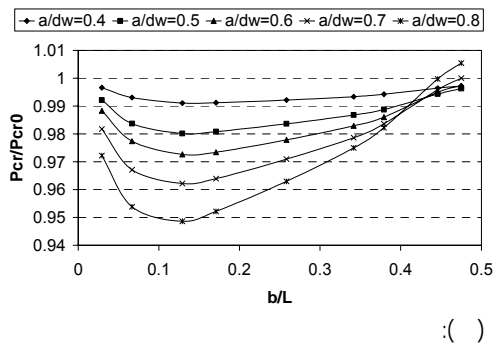
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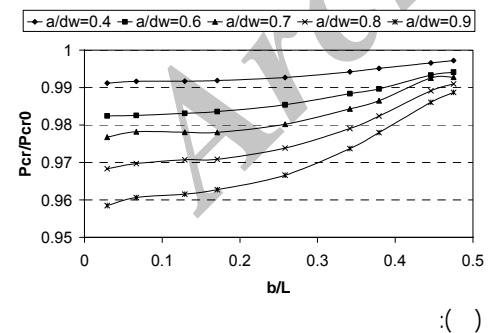
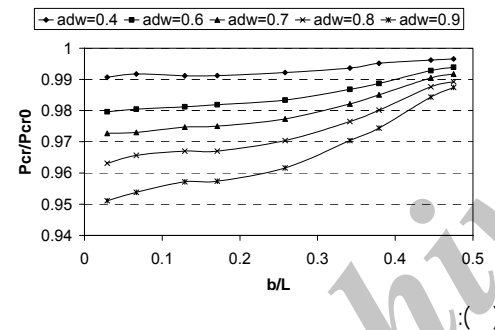
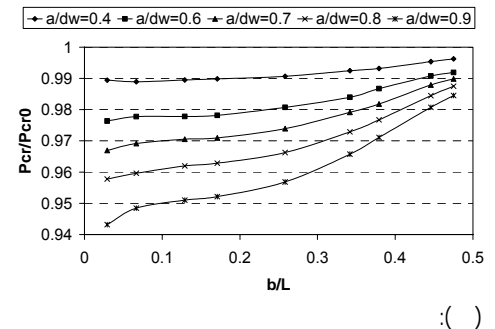
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a

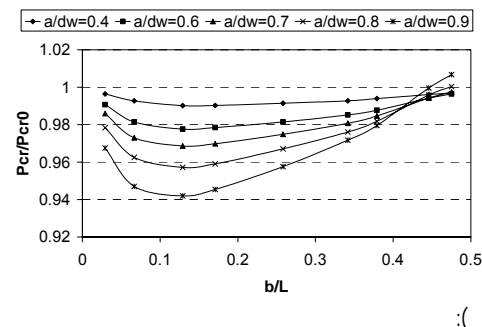
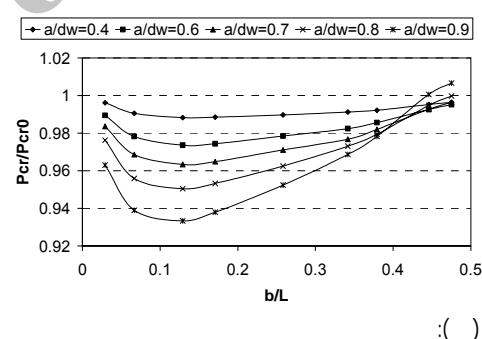
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 $a/d_w = /$   
 $a/d_w = /$   
 $a/d_w = /$   
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