## **Experimental Investigation of Force Coefficients for Groups of Three and Four Circular Cylinders Subjected to a Cross-flow**

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## **ABSTRACT**

This paper has investigated the flow interference between three circular cylinders of equal diameter in an equilateral-triangular arrangement and also between four circular cylinders in a square arrangement when subjected to a cross-flow. Wind tunnel experiments were conducted to measure force coefficients for six spacing ratios ( $l/d$ ) varying from 1.5 to 4 at subcritical Reynolds number of  $6.08 \times 10^4$ . The pressure distributions on the surface of the cylinders were measured, using pressure transducers. It was found that for three cylinders at  $l/d > 2$ , the upstream cylinder experiences lower mean drag coefficient than that of the downstream ones. Also, the minimum drag coefficient values of the downstream cylinders occur at  $l/d = 1.5$  and  $l/d = 2$ . Moreover, It was revealed that for four cylinders at  $l/d = 1.5$ , due to severe flow interference between cylinders, there is a difference between lift coefficients for the upstream cylinders. Also, for  $1/d \le 2$ , the mean drag coefficients for downstream cylinders are negative. In addition, it was concluded that the variations in  $l/d$  strongly affect the aerodynamic coefficients. Also, by decreasing  $l/d$ , the effects of the flow interference between the cylinders increase.

**Key Words:** Multi-cylinder Arrays, Circular Cylinders, Flow Interference, Pressure Distribution

# بررسی تجربی ضرایب نیرو برای گروه های سه و چها*ر* تایی از

## استوانه های دایروی قرا*ر* گرفته در جریان عرضی

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## **چکیده**

در این مقاله، تداخل جریان بین سه استوانه دایروی هم قطر در چیدمان مثلثی و همچنین بین چهار استوانه دایروی در چیدمان مربعی که در جریان عرضی قرار گرفته اند، مورد مطالعه قرار گرفته اند. آزمایشات تونل باد، شامل اندازه گیری ضرایب نیرو برای شش نسبت فضادهی (  $l/d$ ) از 1.5 تا 4 در عدد رینولدز زیر بحرانی  $10^4$   $\times10^4$  می باشد. توزیع فشار روی سطح استوانه ها بوسیله ترانسدیوسرهای فشار اندازه گیری شده است. مشاهده می شود که برای حالت سه استوانه در  $2 > 1$ ، استوانه بالادست ضریب درگ میانگین کمتری را از استوانه های پایین دست متحمل می شود. همچنین، کمترین مقادیر ضرایب درگ استوانه های پایین دست در و 1 = 1 / d قفاق می افتد. بعلاوه، برای حالت چهار استوانه در 1.5 = l/d، به علت تداخل شدید جریان بین / d 1 استوانهها، بین ضرایب لیفت استوانه های بالادست اختلاف وجود دارد. همچنین، برای  $2\leq l/d\leq l$ ، ضرایب درگ میانگین استوانه های پایین دست منفی می شوند. در پایان، نتیجه می شود که تغییرات در  $\,$  ، شدیداً روی ضرایب آیرودینامیکی تأثیر می گذارد و با کاهش اثرات تداخل جريان بين استوانه ها افزايش مى يابد.  ${\it l}/d$ 

**واژه های کلیدی:** آرایه های چند سیلندری، استوانه های دایروی، تداخل جریان، توزیع فشا*د* 

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alone and in groups in the designs for heat exchangers tubes, offshore structures, skyscrapers, many of these engineering applications, fluid the structures. It is possible to model the flows passing through these kinds of structures by multi-

on each cylinder in multi-cylinder arrays are  $\overrightarrow{A}$  three-cylinder array in an equilateral-triangular basically due to vortex shedding quality and characteristics of the flow pattern around the cylinders. When the flow interfere between the cylinders, vortex shedding from each cylinder, vortex shedding on the downstream cylinders and these cases, due to flow interference between cylinders, the fluid usually shows a complicated and surprising behavior. Major points in multiple cylinder arrays studies are to find the effects of the flow interference on each cylinder, fluid-structure interactions and classification of flow pattern around cylinders; whereas, less well studied and

flow, which is a classical problem in fluid mechanics, due to very common and multiple usages, has been studied widely in experimental and numerical methods [6-7]; whereas, most of cases, there have been relatively few numerical studies on the circular cylinders which mostly are based on just two cylinders at low Reynolds

interference between circular cylinders are on two circular cylinders in tandem, side-by-side and staggered arrangements [14-18]. Some studies have investigated the flow around three circular cylinders in an in-line arrangement [19-21]. Sayers [22] studied the drag and lift coefficients occurring on one cylinder in a group of three equispaced cylinders. He concluded that the angle of orientation to the free stream will strongly influence the force coefficients acting on each cylinder. Price and Paidoussis [23] investigated the mean lift and drag forces acting on two and three cylinders in

**1. Introduction** staggered configurations over a wide range of *l* / *d* Cylinder-like structures can be often found both and suggested that the principle of superposition exchangers tubes, offshore structures, skyscrapers,<br>
chemical reaction towers, chimneys, power lines,<br>
the close relationship between aerodynamic cooling systems for nuclear power plants, etc. In responses and pressure distributions on three could apply in determination of force coefficients. Moreover, Yunseok and Changkoon [24] reported the close relationship between aerodynamic circular cylinders.

forces, Strouhal frequencies and flow Despite of numerous works on two cylinders configurations are major criteria for the design of arrangements, only a few studies have been cylinder arrays in a cross-flow. The multi-cylinders many of them were on multi-The steady and fluctuating fluid forces acting cylinders in line, side-by-side or in tandem [25-26]. characteristics of the flow pattern around the arrangement) is a basic unit in many multi-cylinder also near-wake, lead to high fluctuating forces on especially notable in shell and tube heat each cylinder [1]. The effects of mutual interference exchangers. These have also many practical of shear layers, vortices and Karman vortex streets applications in offshore structures. Tatsuno et al. play a vital role in the flow interference  $[2-5]$ . In  $[8]$  investigated the effects of the flow interference Despite of numerous works on two cylinders reported on multi-cylinder arrays such as three and four cylinders arrays. Even among the studies on cylinders in line, side-by-side or in tandem [25-26]. A three-cylinder array in an equilateral-triangular arrangement (or a four-cylinder array in a square arrays such as tube banks. The study of these arrays would be helpful to understand more complicated flow in much larger cylinder arrays, which are between the three cylinders in an equidistant triangular arrangement in a cross-flow. They found that the flow interference effects between three cylinders are severe in the case of low spacing ratio  $(l/d)$  and the flow interference effects weakens

understood have been reported the nature of the identified four different affected regions i.e. small, flow around multiple-cylinder arrays. transition, medium and large spacing ratios for the The flow around a circular cylinder in a cross-flow pattern on three cylinders in an equilateralprevious works have studied multi-cylinder arrays hand, Sayers [27-28] studied the flow interference experimentally in high subcritical Reynolds between four circular cylinders in a square numbers for industrial applications [8-10]. On the arrangement and revealed that the magnitude of the other hand, due to complexity of the flow in these mean lift and drag coefficients are strongly numbers [11-13]. Coefficients of four cylinders in square Many experimental studies on flow configuration for different  $l/d$  and incidence with higher  $l/d$ . In addition, Gu and Sun [9] triangular arrangement at  $Re = 1.4 \times 10^4$  with regards to different levels of interference of influence of spacing ratio  $(l/d)$ . On the other influenced by the orientation of array group to the free stream. Moreover, Lam et al. [10] have the mean and fluctuating force angles ( $\alpha$ ) at subcritical Reynolds numbers, using a piezo-electric load cell.

The present work has experimentally investigated force coefficients for groups of three and four circular cylinders subjected to a cross-flow with six spacing ratios  $(l/d)$  ranging from 1.5 to 4

at subcritical Reynolds number of  $6.08 \times 10^4$ . It is  $6.08\times 10^4$  . It is hoped that this study could provide a useful database and a better understanding of the flow

arrays in a cross-flow.

The experiments were conducted in low-speed, open-circuit wind tunnel of department of  $\overbrace{A}^{\text{flow}}$   $\overbrace{A}^{\text{row}}$ aerospace engineering, K.N. Toosi University of Technology. The wind tunnel test-section is  $1.2 \text{ m}$ wide, 1m high and 3 m long. The maximum velocity in the test-section is 60 m/s. under uniform flow conditions, the longitudinal free-stream turbulence intensity is less than 0.15% and the velocity non-uniformity across the test-section is  $\pm 0.5\%$ .

Figure **1** shows a schematic diagram of the experimental setup in the wind tunnel. The configurations of three cylinders in an equilateraltriangular arrangement and four cylinders in a<br>square arrangement are shown in Fig's 2 and 3. Six Flow square arrangement are shown in Fig's. **2** and **3**. Six  $\qquad \qquad \text{Flow}$ spacing ratios  $(l/d)$  vary from 1.5 to 4, where d is the cylinder diameter,  $l$  is the distance between the center of the adjacent cylinders and  $l/d$  is the spacing ratio between two cylinders. In Fig. **4**, the position of a point on the surface of each cylinder is defined by the azimuthal angle  $\theta$ , measured from the direction of the free-stream flow. Each cylinder of the group is a 92 cm long hollow aluminum tube of 38 mm external diameter, with a machinefinished surface. Thirty pressure taps of diameter<br>0.6 mm are provided overy  $12^{\circ}$  in two perallel rows 0.6 mm are provided every 12° in two parallel rows Flow  $\bigcap_{n=1}^{\infty}$   $\bigcap_{n=1}^{\infty}$   $\bigcap_{n=1}^{\infty}$ within 2 cm distance from each other around mid span in a zigzag manner of each cylinder circumferentially. The aspect ratio is about 24 and the blocking ratio is 2.9% per cylinder. In addition, the total blockage ratio range for the three cylinders is about 7.2–8.7%. Also, the total blockage ratio of the four cylinders is 5.8%. All tests were carried out at the Reynolds



**Fig. (1):** Schematic diagram of the experimental



**Fig. (2):** The configuration of three circular cylinders in an equilateral-triangular arrangement.



**Fig. (3):** The configuration of four circular cylinders in a square arrangement.



**Fig. (4):** Angel  $\theta$ : the position of a point on the surface of each cylinder.

number of  $6.08 \times 10^4$ , based on the diameter of a single cylinder and a free-stream velocity of 24 m/s. During the experiments, reference flow conditions were measured with a Pitot-static tube and a micro manometer. The measurement system of the surface pressure was consisted of pressure transducers (Honeywell-DC005NDC4), a National Instruments (NI) PCI-6224 16-bit A/D board with 32 analogue input channels and a personal computer.

setup in the wind tunnel. In this study, the pressure coefficient  $C_p$ , the lift The estimated measurement uncertainties of the pressure, lift and drag coefficients are  $C_p \pm 0.01$ ,  $C_L \pm 0.015$  and  $C_D \pm 0.02$ , respectively.  $C_p \pm 0.01$ ,  $C_L \pm 0.015$  and  $C_D \pm 0.02$ , respectively.<br>In this study, the pressure coefficient  $C_p$ , the lift , the lift coefficient  $C_L$  and the drag coefficient  $C_D$  for each cylinder are defined as and the drag coefficient  $C_D$  for for each cylinder are defined as  $C_P = (P - P_{\infty})/(0.5 \rho V_{\infty}^{2})$  $P-P_{\infty}$   $\frac{1}{0.5\rho V_{\infty}^{2}}$ ,

$$
C_L = L/(0.5 \rho V_\infty^2 d) \& C_D = D/(0.5 \rho V_\infty^2 d)
$$
  
respectively, where *P* is the mean static pressure  
on the surface of the cylinder,  $P_\infty$ , the static  
pressure of the free-stream flow, *L*, the lift force,

*D*, the drag force,  $\rho$ , the air density,  $V_{\infty}$  and the  $\sigma$ free-stream velocity.

The pressure distributions on the surface of the cylinders were measured by the pressure coefficient on the surface of a single cylinder is  $\theta$  (deg) presented. In this study, the pressure distribution method has been employed for measuring mean force coefficients in which lift and drag coefficients are only calculated by pressure integration around the cylinders at mid-span. As depicted in Fig. **5**, for a single cylinder at  $Re = 6.08 \times 10^4$ , it is found that  $C_L = 0$  and  $C_D = 1.20$ , which shows a  $\bigcup_{\text{--cyl. A}}$ good agreement with the results reported by Lam et  $\Big\|_{0.5}$   $\Big\|$   $\Big\$ al. [10] for a single cylinder which has been measured at  $Re = 6 \times 10^4$  using a piezo-electric load cell ( $C_D = 1.217$  and  $C_L = 0$ ).  $\frac{8}{9}$ 



**Fig. (5):** The pressure coefficient distribution on the  $\left\{\begin{matrix} \end{matrix}\right\}$   $\left\{\begin{matrix} \end{$ surface of a single cylinder.

Figures **6** to **11** show the pressure coefficient distributions on the surface of the three circular cylinders in equilateral-triangular arrangement for different values of  $l/d = 1.5$  to 4. In addition, Figs. **12** and **13** show variations of the lift and drag



**Fig. (6):** The pressure coefficient distributions of the three cylinders for  $l/d = 1.5$ .



**Fig. (7):** The pressure coefficient distributions of



**Fig. (8):** The pressure coefficient distributions of the three cylinders for  $l/d = 2.5$ .



**Fig. (9):** The pressure coefficient distributions of the three cylinders for  $l/d = 3$ .



**Fig. (10):** The pressure coefficient distributions of



**Fig. (11):** The pressure coefficient distributions of



with  $l/d$  (for three cylinders).



with  $l/d$  (for three cylinders).

the three cylinders for  $l/d = 3.5$ . there is an attractive force between the two <sup>2</sup>
<sup>2</sup>
coefficients rapidly increase. This is a general trend **-1.5 -1** minimum drag coefficient values for the  $\text{experiences lower mean drag force } (C_D)$  than that hand, for  $l/d > 2$ , the upstream cylinder  $\begin{array}{c} 0.5 \rightarrow \infty. \end{array}$  Cyl. B  $\begin{array}{c} \rightarrow \infty. \end{array}$  considerably which indicates substantial changes in <sup>1</sup> **0 60 120 180 240 300 360** Cyl. B **A** Considerably which indicates substantial changes in **Cylensis** Cylinders. On the other than the flow pattern around the cylinders. On the other In Fig. **12**, it is evident that when three cylinders are in equilateral-triangular arrangement, cylinders B and C. Furthermore, according to the values of  $C_L$  for the downstream cylinders, it is for the downstream cylinders, it is clear that for slight variations of  $l/d$ , lift hand, for  $l/d > 2$ , the upstream cylinder ) than that of the downstream cylinders (see Fig. **13**). The downstream cylinders occur at  $l/d = 1.5$  and  $l/d = 2$ , but for  $l/d$  higher than 2 the drag that by increasing  $l/d$ , mean drag coefficients for all cylinders have almost been increased.

the three cylinders for  $l/d = 4$ . According to Fig. 12, it is revealed that for  $l/d = 1.5$ , there is a difference between lift

coefficients for two cylinders B and C in side-by side arrangement as a result of severe flow interference between the cylinders. The same  $\bigcup_{\text{subsub}} \bigcup_{\text{subsub}} A$   $\longrightarrow$  cylinders. phenomenon has been found for three cylinders in  $0.5 \rightarrow \rightarrow \text{Cu}$ . Cylinders in an equilateral-triangular arrangement which has  $\begin{bmatrix} \uparrow \\ \uparrow \end{bmatrix}$ been reported by Tatsuno et al. [8]. From the other viewpoint, by increasing the distance between the  $\frac{d}{d}$ cylinders for  $l/d = 4$  and so weakening the flow interference between the cylinders, it is found that the lift and drag coefficients for cylinder A is almost similar to that of a single cylinder. Thus, it is concluded that the variations in  $l/d$ , strongly affect the aerodynamic coefficients and also by  $\begin{array}{cccc} a & b & b & b \\ b & c & b & 120 \\ d & c & d & 240 \end{array}$  are  $\begin{array}{cccc} a & b & b \\ b & c & d \end{array}$ decreasing the  $l/d$ , the effects of the flow

## **3.2. Four Cylinders**

Figures **14**-**19** show the pressure coefficient distributions on the surface of the four circular cylinders in square arrangement for different values of  $l/d = 1.5$  to 4. Moreover, Figs. 20-21 show  $\left\{\begin{matrix} 0 \\ 1 \end{matrix}\right\}$   $\left\{\begin{matrix} 0 \\ 1 \end{matrix}\right\}$ variations of the lift and drag coefficients with  $0.5 \begin{array}{c} \downarrow \\ \downarrow \end{array}$ respect to  $l/d$  at  $Re = 6.08 \times 10^4$ .



**Fig. (14):** The pressure coefficient distributions of



**Fig. (15):** The pressure coefficient distributions of the four cylinders for  $l/d = 2$ .



interference between the cylinders increase.<br>the four cylinders for  $l/d = 2.5$ . **Fig. (16):** The pressure coefficient distributions of the four cylinders for  $l/d = 2.5$ .



**-1.5<sup>4</sup> / 1.5 Fig. (17):** The pressure coefficient distributions of the four cylinders for  $l/d = 3$ .



the four cylinders for  $l/d = 3.5$ .





**Fig. (20):** Variation of the mean lift coefficients



In Fig. **20**, it is evident that when four cylinders are in square arrangement, there is a repulsive force between the upstream cylinders A and B which is weakened by increasing  $l/d$ . Furthermore, it can

**-0.5**  $\sqrt{a^2}$  occur at  $l/d = 3$ . By increasing  $l/d$ , the mean  $\begin{array}{c} 0.5 \rightarrow \end{array}$  **Cyles**<br>  $\begin{array}{c} \uparrow \\ \text{+ Cyl. c} \end{array}$  that of the upstream cylinders and the minimum **1 Cyl. B Cyl. D**  $\left| \right|$  drag coefficient values for the upstream cylinders experience lower mean drag coefficients ( $C_D$ ) than ) than drag coefficients for downstream cylinders are increased. It is notable that for  $l/d \leq 2$  these coefficients are negative.

 $\epsilon/\omega$  and  $\epsilon$  is a directive set of the set of  $\epsilon$ **0 60 120 180 240 300 360** coefficients for the upstream cylinders A and B as a **Fig. (19):** The pressure coefficient distributions of cylinders. On the other hand, by increasing the the four cylinders for  $l/d = 4$ . distance between the cylinders for  $l/d = 4$  and  $l/d$ , the effects of the flow interference between **0.1 Cyl. C Cyl. B** similar to that of a single cylinder. So, it is found  $\mathcal{L}_{\mathsf{cyl},\mathsf{D}}$  that the variations in  $l/d$ , strongly affect the  $l/d = 1.5$ , there is a difference between lift result of severe flow interference between the thus weakening the flow interference between the cylinders, it is revealed that the lift and drag the cylinders increase.

### **4. Conclusions**

**1 1.5 2 2.5 3 3.5 4 4.5** arrangement was investigated when subjected to a with  $l/d$  (for four cylinders). measured experimentally at subcritical Reynolds Flow interference between three circular cylinders cross-flow. For this purpose, force coefficients for six spacing ratios  $(l/d)$  varying from 1.5 to 4 were number of  $6.08 \times 10^4$ . Some useful conclusions are obtained and they are summarized as follows:

- **0.4**  $\begin{array}{c} 0.8 \rightarrow \text{Cyl} \$ 1.2 **exert attractive force on each other.** Moreover, **Cyl. A** Cyle **B Cyllengers Cyllengers**, there is a difference between **Cyl. C** lift coefficients for downstream cylinders B and **Cyl. D** current configuration, two cylinders B and C for  $l/d = 1.5$ , due to severe flow interference C. The contract of the contrac
- $\begin{array}{ccccccccc}\n\text{--}\n\text{--}\n\end{array}$   $\begin{array}{ccccccccc}\n\text{--}\n\text{--}\n\end{array}$   $\begin{array}{ccccccccc}\n\text{--}\n\text{--}\n\end{array}$  downstream cylinders. Also, the minimum drag  $l/d > 2$ , the upstream cylinder experiences **1 1.5 2 2.5 3 3.5 4 4.5 Fig. (21):** Variation of the mean drag coefficients  $l/d = 1.5$  and  $l/d = 2$ . Moreover, by with  $l/d$  (for four cylinders). increasing  $l/d$ , mean drag coefficients for all 2-In the case of three cylinders, it is found that for lower mean drag coefficient than that of the downstream cylinders. Also, the minimum drag coefficients of the downstream cylinders occur at  $l/d = 1.5$  and  $l/d = 2$ . Moreover, by cylinders, nearly increase.
	- 3-When the four cylinders are arranged in the current configuration, upstream cylinders A and B exert repulsive force on each other which is

weakening by increasing  $l/d$ . In addition, for  $l/d = 1.5$ , due to severe flow interference 2003.

- 4-In the case of four cylinders, the downstream Fluids Mech., Vol. 105, No. 2, pp. 397-425, 1981. cylinders experience lower mean drag for the upstream cylinders occur at  $l/d = 3$ . On the other hand, by increasing  $l/d$ , the mean drag coefficients for downstream cylinders are  $\overline{X}$  are  $\overline{X}$  and  $\overline{X}$  and  $\overline{X}$  and  $\overline{X}$  and  $\overline{X}$  and  $\overline{X}$
- 5-It is revealed that the effects of variations in  $l/d$ ,<br>on the aerodynamic coefficients are strong and<br> $\frac{315,1998}{9 \text{ Gu. } 7 \text{ E and Sun. } \Gamma \text{ E}$  "Classification of Flow three cylinders, at  $l/d = 4$  flow behavior 568, 2001. around cylinder A to be somehow the same as that of around a single cylinder. Also, for four cylinders, at  $l/d = 4$  flow behavior around each

should focus on the effects of Reynolds number in regimes, high free-stream turbulence intensity, incidence angle, presence of thermal gradients dimensional cylinders with finite span, on the flow pattern and the aerodynamic coefficients, when the flow interfere between three or four circular

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